



## **PART 21**

CAA Consolidation

### **Certification of Products and Parts**

## **DESCRIPTION**

Part 21 prescribes the requirements for –

- the type certification of a product to be manufactured in Mongolia,
- the type acceptance certification of a product to be imported into Mongolia,
- the airworthiness certification of an aircraft,
- the approval or authorisation of a design change, material, part, process, appliance, technical data, and a critical part,
- the issue of a document for the export of a product, critical part, or appliance, and
- the identification of a product or a critical part.

**This document is the current consolidated version of Part 21 produced by the Civil Aviation Authority, and serves as a reference only. It is compiled from the official ordinary rules that have been signed into law by the Minister of Road and Transport. Copies of the official rule and amendments as signed by Minister may be obtained from the Civil Aviation Authority or may be downloaded from the official web site at: [www.mcaa.gov.mn](http://www.mcaa.gov.mn)**

**Bulletin**

ICAO 29<sup>th</sup> Assembly Resolution A29-3 of year 1992 urges States to promote global harmonization of national rules.

In order to implement this Resolution, Mongolian Civil Aviation Safety Regulation has been developed based on “Memorandum for Technical Cooperation” between CAA of Mongolia and New Zealand, signed on 6<sup>th</sup> of May, 1999.

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## Subpart A — General

### 21.1. Purpose

This Part prescribes rules governing—

- (1) the type certification of a product to be manufactured in Mongolia:
- (2) the type acceptance certification of a product to be imported into Mongolia:
- (3) the airworthiness certification of an aircraft:
- (4) the approval or authorisation of a design change, material, part, process, appliance, technical data, and a critical part:
- (5) the issue of a document for the export of a product, critical part, or appliance:
- (6) the identification of a product or a critical part.

### 21.3. Definitions

In this Part:

**Type certificate** includes

- (1) the type design; and
- (2) the operating limitations; and
- (3) the type certificate data sheet; and
- (4) the applicable airworthiness design standards specified in Appendix C; and
- (5) for an aircraft type, the flight manual; and
- (6) any other conditions or limitations prescribed for the product type under this Part.

**Critical part** means an aircraft part for which a replacement time, inspection interval, or related procedure is specified in the Airworthiness Limitations section of a manufacturer's Maintenance Manual or Instructions for Continued Airworthiness.

### 21.5. Reserved

### 21.7. Foreign applications

Applications for certificates, approvals, or authorisations from persons located outside of Mongolia shall only be considered for approval if the Director is satisfied that there is a need for the certificate, approval, or authorisation.



## 21.8. Required design changes

(a) Where the Director issues an airworthiness directive for a product under Part 39, the holder of the type certificate for the product type shall—

- (1) if the Director determines that design changes are necessary to correct the unsafe condition of the product, upon the Director's request, submit appropriate design changes to the Director for approval; and
- (2) upon approval of the design changes, make the descriptive data covering the changes available to all operators of the product.

(b) If there are no current unsafe conditions of the product but it is found through service experience that changes in the type certificate will contribute to the safety of the product, the holder of the type certificate for a product type shall—

- (1) upon the Director's request, submit appropriate design changes to the Director for approval; and
- (2) upon approval of the design changes, make the descriptive data covering the changes available to all operators of the product.

## Subpart B — Type Certificates and Type Acceptance Certificates

### 21.11. Purpose

This Subpart prescribes rules governing—

- (1) the type certification of a product to be manufactured in Mongolia;
- (2) the type acceptance certification of a product to be imported into Mongolia;
- (3) the responsibilities of a holder of a type certificate.

### 21.13. Certificate categories

(a) The following certificates may be issued by the Director under this Subpart:

- (1) *standard category* type certificate for a product type to be manufactured in Mongolia;
- (2) *restricted category* type certificate for an aircraft type to be manufactured in Mongolia;
- (3) *standard category* type acceptance certificate for a product type to be imported into Mongolia;
- (4) *restricted category* type acceptance certificate for an aircraft type to be imported into Mongolia.

(b) The Director may issue a certificate specified in paragraph (a) if the applicable requirements for the certificate are met.

(c) When issuing a *restricted category* type certificate or a *restricted category* type acceptance certificate for an aircraft, the Director must specify in the certificate the operational purposes for which the aircraft is certificated.

#### **21.15. Reserved**

#### **21.17. Application for certificate**

(a) An applicant for the issue of a type certificate for a product type must complete form CAA 24021/01 and submit it to the Director with—

- (1) a payment of the applicable fee prescribed by regulations made under the Act; and
- (2) the following information—
  - (i) for an aircraft type, a three-view drawing of the aircraft type and any available preliminary basic data; and
  - (ii) for an aircraft engine or propeller type, a description of its design features, its operating characteristics, and its proposed operating limitations; and
  - (iii) any further particulars relating to the product as required by this Subpart or may be required by the Director.

(b) An application for the issue of a type certificate for a product type remains in effect for a period of—

- (1) 3 years; or
- (2) 5 years for an aeroplane type with a MCTOW greater than 5700 kg; or
- (3) 5 years for a rotorcraft type with a MCTOW greater than 2730 kg; or
- (4) a longer period that may be approved by the Director.

(c) An applicant for the issue of a type acceptance certificate for a product type must complete form CAA 24021/02 and submit it to the Director with a payment of the applicable fee prescribed by regulations made under the Act.

#### **21.19. Issue of certificate**

(a) An applicant is entitled to the issue of a type certificate or a type acceptance certificate for a product type if the Director is satisfied that—

- (1) the applicant meets the applicable certification requirements in rules 21.31 to 21.43; and

- (2) the issue of the certificate is not contrary to the interests of aviation safety; and
  - (3) for a type certificate for a product type—
    - (i) the inspections and tests required under rule 21.35 confirm that the product type meets the applicable airworthiness requirements; and either
    - (ii) the applicant holds or has applied for a *design organisation certificate* in accordance with Part 146; or
    - (iii) the applicant holds or has applied for a *manufacturing organisation certificate* in accordance with Part 148
- (b) If a type certificate is not issued within the time period specified in rule 21.17(b) for the application, the applicant may, in accordance with rule 21.17(a),
- (1) submit a new application for a type certificate; or
  - (2) submit an application to extend the original application.
- (c) If an application submitted under paragraph (b)(2) to extend an application for a type certificate is accepted, the effective date of the original application is deemed to be a date that precedes the actual date of issue of the type certificate by a period not greater than the period established under rule 21.17(b) for the original application.

#### **21.21. Reserved**

#### **21.23. Special conditions**

The Director may prescribe special conditions for a product to establish a level of safety equivalent to the airworthiness design standards specified in Appendix C if the Director determines that the airworthiness standards do not contain adequate or appropriate safety levels because—

- (1) the product has novel or unusual design features relative to the design practices on which the applicable airworthiness design standards are based; or
- (2) the intended use of the product is unconventional.

#### **21.25. Duration of certificate**

- (a) A type certificate remains in force unless the certificate is suspended or revoked.
- (b) A type acceptance certificate remains in force unless the type certificate or other equivalent document required under rule 21.43(a)(1)(i) for the issue of the type acceptance certificate is no longer valid.

### **21.27. Transfer of certificate**

(a) A holder of a type certificate must not transfer the certificate to another person unless the person to hold the type certificate is acceptable to the Director and the person holds—

- (1) a *design organisation certificate* issued in accordance with Part 146; or
- (2) a *manufacturing organisation certificate* issued in accordance with Part 148.

(b) A holder of a type certificate must, before transferring a type certificate,—

- (1) notify the Director in writing of the name and address for service of the transferee; and
- (2) produce the type certificate to the Director for amendment.

## **Type Certification Requirements**

### **21.31. Airworthiness requirements**

An applicant for the grant of a type certificate for a product type must provide the Director with evidence that—

- (1) except as provided in paragraph (2), the product type design complies with—
  - (i) the applicable airworthiness design standard specified in Appendix C or any later amendment of that standard as selected by the applicant; and
  - (ii) any special conditions prescribed by the Director under 21.23; and
  - (iii) any other airworthiness requirement the Director considers to be relevant; and
  - (iv) the applicable aircraft noise and engine emission standards specified in Appendix C or any later amendment of those standards as selected by the applicant; and
- (2) any airworthiness requirement not complied with is compensated for by factors providing an equivalent level of safety; and
- (3) no feature or characteristic of the product type makes it, when operated in accordance with the correctly amended flight manual or other prescribed limitations, unsafe for the intended use.

### **21.32. Aircraft noise and engine emission certification**

An applicant for the grant of a type certificate for a product type must provide the Director with—

- (1) for an aircraft type, aircraft noise certification information consisting of—

- 
- (i) the manufacturer's type and model designation, engine type and model, and if applicable the propeller type and model; and
  - (ii) a statement identifying the applicable noise standard for the aircraft type in accordance with the standard category aircraft noise standard specified in Appendix C; and
  - (iii) a list of any modifications that have been incorporated in the aircraft design to ensure compliance with the noise standard identified by the applicant under paragraph (1)(ii); and
  - (iv) a statement specifying the average aircraft noise levels measured in accordance with the standards and procedures prescribed for the noise standard identified by the applicant under paragraph (1)(ii); and
  - (v) a statement specifying the maximum weight of the aircraft during the noise level measurements required in paragraph (1)(iv); and
- (2) for turbojet and turbofan engine types, engine emission certification information meeting the standard category engine emission standards specified in Appendix C, including —
- (i) the manufacturer's type and model designation; and
  - (ii) a statement identifying the applicable engine emission standard for the engine type, in accordance with the standard category aircraft engine emission standard specified in Appendix C; and
  - (iii) a list of any modifications that have been incorporated in the engines to ensure compliance with the emission standard identified by the applicant under paragraph (2)(ii); and
  - (iv) a statement of the rated output as prescribed in the emission standard identified by the applicant under paragraph (2)(ii); and
  - (v) a statement of the reference pressure ratio as required by the emission standard identified by the applicant under paragraph (2)(ii); and
  - (vi) a statement identifying compliance with the Smoke Number requirements in the emission standard identified by the applicant under paragraph (2)(ii); and
  - (vii) a statement identifying compliance with the gaseous pollutant requirements in the emission standard identified by the applicant under paragraph (2)(ii).

**21.33. Type design**

An applicant for the grant of a type certificate for a product type must—

- (1) provide the Director with a type design that consists of—
  - (i) the drawings, specifications, and information necessary to define the configuration and the design features of the product type which have been shown to comply with the applicable airworthiness requirements; and
  - (ii) a list of those drawings and specifications provided under paragraph (1)(i); and
  - (iii) the information on dimensions, materials, and processes necessary to define the structural strength of the product type; and
  - (iv) the Airworthiness Limitations section of the Instructions for Continued Airworthiness, required in the applicable airworthiness design standards specified in Appendix C; and
  - (v) any other data necessary to allow the determination of the airworthiness of later products of the same type; and
- (2) identify each type design and each variant within the type design.

**21.35. Inspections and tests**

(a) Each applicant for the grant of a type certificate for a product type shall inspect and test a product of the type to ensure that—

- (1) the product complies with the applicable airworthiness requirements; and
- (2) the materials and product conform to the specifications in the type design; and
- (3) all parts of the product conform to the drawings in the type design; and
- (4) the manufacturing processes, construction and assembly conform to those specified in the type design.

(b) The applicant shall, after making the inspections and tests required under paragraph (a)—

- (1) permit the Director to perform any inspection and flight and ground testing that the Director may require; and
- (2) provide evidence that the product meets the requirements in paragraphs (a)(2), (3) and (4); and
- (3) ensure that the product remains unchanged between the time that the product is shown to meet the requirements in paragraphs (a)(2), (3) and (4) and presentation to the Director for testing.

**21.37. Statement of compliance**

(a) Each applicant for the grant of a type certificate presenting a product to the Director for tests under 21.35(b) shall provide the Director with a statement of compliance stating that the applicant has complied with the requirements of 21.35(b)(2) and (3).

(b) Each applicant for the grant of a type certificate for a product type shall provide the Director with a statement of compliance stating that the product complies with the applicable airworthiness requirements.

**21.39. Flight tests**

(a) Subject to paragraph (b), an applicant for the issue of a type certificate for an aircraft type must make a flight test that the Director may require to verify that—

- (1) the aircraft type complies with the applicable airworthiness requirements; and
- (2) the aircraft type and the aircraft components are reliable and function properly.

(b) The applicant must ensure that—

- (1) before a flight test that is required under paragraph (a) is made—
  - (i) the aircraft complies with the structural requirements of the applicable airworthiness design standards specified in Appendix C; and
  - (ii) the aircraft has undergone the necessary ground inspections and tests; and
  - (iii) the aircraft conforms to the type design; and
- (2) every flight test that is required under paragraph (a) is conducted—
  - (i) by a person who meets the requirements prescribed in below:
    - (A) the person holds a valid pilot licence or validation permit issued in accordance with Part 61 and is approved by the Director to act as a test pilot for the type of prototype testing or experimental flying that is being undertaken; or
    - (B) the aircraft is a microlight, and the person is authorised by a microlight organisation for the purpose if the organisation's Part 149 certificate authorises the holder to approve a test pilot for the type of prototype testing or experimental flying that is being undertaken; or
    - (C) the aircraft is a glider, and the person is authorised by a gliding organisation for the purpose if the organisation's Part 149 certificate authorises the holder to approve a test pilot for the type of prototype testing or experimental flying that is being undertaken;

- (ii) in accordance with applicable procedures that are acceptable to the Director.

## **Type Acceptance Certification Requirements**

### **21.41. Airworthiness requirements**

An applicant for the issue of a type acceptance certificate for a product type must provide the Director with evidence that—

- (1) the product type meets the applicable —
  - (i) airworthiness design standards; and
  - (ii) aircraft noise and engine emission standards specified in Appendix C — and effective at the date assigned in the foreign type certificate or an equivalent document, unless another date is specified in writing by the Director; and
- (2) the product type meets any special conditions imposed under the foreign type certification or as specified in writing by the Director under rule 21.23; and
- (3) any airworthiness requirement not complied with is compensated for by a factor that provides an equivalent level of safety; and
- (4) there is no feature or characteristic of the product type that makes it unsafe for the intended use.

### **21.43. Data requirements**

(a) An applicant for the issue of a type acceptance certificate for a product type must provide the Director with—

- (1) evidence that—
  - (i) the type design has been approved by an ICAO contracting State of Design by the issue of a type certificate or an equivalent document; and
  - (ii) the type certificate or equivalent document is considered to be valid by the State of Design; and
- (2) details of the airworthiness requirements complied with, for the issue of the type certificate or equivalent document specified in paragraph (a)(1), including—
  - (i) the airworthiness design standards; and
  - (ii) the effective date of the standards; and
  - (iii) any special condition imposed under the foreign type certification; and



- (iv) any requirement not complied with and any compensating factor that provides an equivalent level of safety acceptable to the Director; and
  - (v) any airworthiness limitation; and
  - (3) details of the information required in rule 21.32(1) and (2) regarding aircraft noise and engine emission certification; and
  - (4) a list identifying the data submitted for the issue of the type certificate or equivalent document specified in paragraph (a)(1), showing compliance with the applicable airworthiness design standards; and
  - (5) for an aircraft, a copy of the flight manual approved under the type certificate or equivalent document specified in paragraph (a)(1) or, if the applicable design standards do not require a flight manual to be provided, a flight manual meeting the standards prescribed in Appendix C; and
  - (6) if required by the Director,—
    - (i) a copy of the maintenance manual for the product type; and
    - (ii) a copy of the current service information issued by the manufacturer of the product type; and
    - (iii) a copy of an illustrated parts catalogue for the product type; and
  - (7) evidence that the manufacturer, or holder of the type certificate or equivalent document specified in paragraph (a)(1), has agreed to provide the Director with a copy of every amendment and re-issue of the documents prescribed in paragraphs (a)(5) and (a)(6).
- (b) If the product type referred to in paragraph (a) is an aircraft type, the information required under paragraph (a)(6) must include the aircraft engine, and propeller, if applicable.
- (c) The Director may specify in writing the range of serial numbers or models of products to which the type acceptance certificate relates, or redefine the applicability of the certificate if the requirements under rules 21.41 and 21.43 are satisfied for any additional product.

## **Subpart C — Design Changes**

### **21.71. Purpose**

This Subpart prescribes rules governing the approval of a design change.

**21.73. Approval of design changes**

- (a) A design change may be approved by—
- (1) including it in an Airworthiness Directive; or
  - (2) the approval of a modification; or
  - (3) the approval of a change to the type certificate or type acceptance certificate under Subpart D; or
  - (4) the issue of a supplemental type certificate under Subpart E.
- (b) Design changes are acceptable to the Director if they are—
- (1) described by technical data listed in Appendix D; or
  - (2) accepted by the issue of an airworthiness certificate.
- (c) A person who is not the type certificate holder for a product and who intends to make a major modification to that product may apply to the Director for approval of the design change by the issue of a supplemental type certificate under Subpart E.

**21.75. Continuation of design changes**

Each design change that has been approved by the Director at the time this Part comes into force shall be deemed to have been approved under this Part.

**21.77. Continued airworthiness responsibilities**

The organisation under which the statement of compliance for a design change was issued shall undertake the continued airworthiness responsibilities required by 146.61 in respect of the change.

**21.79. Acceptance of design changes by the issue of an airworthiness certificate**

- (a) A design change embodied on an aircraft may be accepted by the issue of an airworthiness certificate for that aircraft if —
- (1) the design change is recorded in the maintenance records for the aircraft at the time of issue of the airworthiness certificate; and
  - (2) there is sufficient evidence that the design change meets the applicable airworthiness requirements; and
  - (3) there is sufficient evidence that the design change meets the applicable aircraft noise and engine emission standards specified in Appendix C.
- (b) A design change accepted under this rule may only be embodied on the individual aircraft for which the airworthiness certificate has been issued.

**21.81. Form CAA 337 – approval of modifications**

The Director shall approve a modification by approving the modification's technical data under 21.505.

**Subpart D — Changes to Type Certificates and Type Acceptance Certificates****21.91. Purpose**

This Subpart prescribes rules governing the approval of a change to a type certificate and a type acceptance certificate.

**21.93. Changes to certificates**

(a) A change to a type certificate in accordance with this Subpart may include a change to any of the following:

- (1) the type certificate category;
- (2) the type design;
- (3) the flight manual;
- (4) the operating limitations;
- (5) the type certificate data sheet;
- (6) any special condition specified in the type certificate.

(b) A change to a type acceptance certificate must be made in accordance with rule 21.95(b).

**21.95. Design changes requiring a new certificate**

(a) Each applicant for a design change approval shall make a new application for a type certificate where—

- (1) the Director determines that the proposed change in the design, configuration, power, power limitations for engines, speed limitations for engines, or weight, is so extensive that a substantially complete investigation of compliance with the airworthiness requirements is required; or
- (2) for an aircraft type, the proposed change is to—
  - (i) the number of engines or rotors; or
  - (ii) engines or rotors using different principles of propulsion; or
  - (iii) rotors using different principles of operation; or
- (3) for an aircraft engine type, the proposed change is in the principle of operation; or

(4) for a propeller type, the proposed change is in—

- (i) the number of blades; or
- (ii) the principle of pitch change operation; or
- (iii) the blade material.

(b) Each applicant for the approval of a change to a type acceptance certificate shall make a new application for a type acceptance certificate in accordance with Subpart B and provide the Director with evidence that the applicable foreign authority has approved the change to its type certificate in accordance with the applicable airworthiness requirements.

#### **21.97. Limitations of changes to type certificates**

An approved change to a type certificate shall be limited to the specific configuration upon which the change is made unless the applicant provides the Director with—

- (1) the identification of the other configurations of the same type for which approval is requested; and
- (2) evidence that the change is compatible with the other configurations.

#### **21.99. Eligibility**

The applicant for a change to a type certificate shall be the certificate holder.

#### **21.101. Application for approval**

Each applicant for a change to a type certificate shall complete form CAA 24021/08, which shall require—

- (1) the name and address for service in Mongolia of the applicant; and
- (2) the information requested in 21.505; and
- (3) any descriptive data for inclusion in the certificate; and
- (4) such further particulars relating to the product and applicant as may be required by the Director as indicated on the form—

and submit it to the Director with a payment of the applicable fee prescribed by regulations made under the Act.

#### **21.103. Approval of change to type certificate**

The Director may approve a change to a type certificate and an applicant is entitled to the approval of a change to a type certificate if—

- (1) the applicant meets the applicable requirements of this Subpart in a manner acceptable to the Director; and
- (2) the change to the type certificate is not contrary to the interests of aviation safety; and

- (3) any airworthiness requirement not complied with is compensated for by a factor that provides a level of safety acceptable to the Director for the intended continuing use of the aircraft; and
- (4) there is no feature or characteristic of the changed product that makes it unsafe for the intended use if it is operated in accordance with the correctly amended flight manual or other specified limitations.

## **Subpart E — Supplemental Type Certificates**

### **21.111. Purpose**

This Subpart prescribes—

- (1) rules governing the issue of a supplemental type certificate; and
- (2) the responsibilities of a holder of a supplemental type certificate.

### **21.113. Supplemental type certificate**

A supplemental type certificate issued in accordance with this Subpart may allow changes of the following type or in the following circumstances:

- (1) to the type certificate category or type acceptance certificate category; or
- (2) to the type design; or
- (3) to the flight manual; or
- (4) to the operating limitations; or
- (5) to any special conditions prescribed on the type certificate or type acceptance certificate; or
- (6) where a person who is not the type certificate holder for a product intends to make a major modification to the product; or
- (7) where a person wishes to obtain a supplemental type certificate for any valid commercial purpose (for example, the applicant is party to a sale to a second party or requires such a certificate for acceptance by a foreign regulatory authority).

### **21.115. Eligibility**

Any person may apply for a supplemental type certificate.

### **21.117. Application for certificate**

An applicant for a supplemental type certificate must complete form CAA 24021/09 and submit it to the Director with —

- (1) the name and address for service in Mongolia of the applicant; and
- (2) the information requested in—

- (i) 21.33; and
  - (ii) 21.35; and
  - (iii) 21.505(a); and
  - (iv) 21.32 if the proposed changes may affect the aircraft noise or engine emission levels; and
- (3) such further particulars relating to the design change and the applicant as may be required by the Director as indicated on the form; and
- (4) payment of the applicable fee prescribed by regulations made under the Act.

#### **21.119. Issue of certificate**

(a) The Director may issue a supplemental type certificate for a product or product type and an applicant is entitled to a supplemental type certificate if the Director is satisfied that—

- (1) the applicant meets the applicable requirements of this Subpart; and
- (2) the issue of the certificate is not contrary to the interests of aviation safety; and
- (3) any airworthiness requirement that is not complied with is compensated for by a factor that provides an equivalent level of safety; and
- (4) there is no feature or characteristic of the changed product that makes it unsafe for the intended use, if the changed product is operated in accordance with the correctly amended flight manual or other specified limitation.

(b) A certificate issued by the Director under this Subpart may be subject to conditions as the Director considers appropriate in each particular case.

#### **21.121. Transfer of certificate**

(a) The holder of a supplemental type certificate shall not transfer the certificate to an organisation or person other than an organisation or person accepted by the Director as able to undertake the responsibilities of 21.123.

(b) The holder of a supplemental type certificate shall, before transferring the certificate—

- (1) notify the Director in writing, of the name and address for service in Mongolia of the transferee; and
- (2) produce the certificate to the Director for amendment.

#### **21.123. Responsibilities of certificate holder**

(a) The holder of the supplemental type certificate shall—

- (1) undertake the continued airworthiness responsibilities required by 146.61 in respect of the change to a type certificated product; and

- (2) upon the Director's request, provide the Director with evidence of appropriate liaison with the holder of the type certificate of the product; and
- (3) ensure that—
  - (i) all records are legible and of a permanent nature; and
  - (ii) except as provided in paragraph (b), the record of a design or design change is retained for a period of 2 years from the date the last example of the product type is permanently withdrawn from service; and
- (4) upon the Director's request, make the certificate, design information, drawings, test reports, and inspection records available to the Director.

(b) The Director may permit records to be retained for a lesser period than that required by paragraph (a)(3)(ii).

### **21.125. Changes to certificate**

The holder of a supplemental type certificate intending to make a significant change to the design change described in the certificate shall—

- (1) apply for an amendment to the supplemental type certificate on form CAA 24021/09; or
- (2) apply for a new supplemental type certificate.

## **Subpart F — Provisional Type Certificates**

### **21.141. Purpose**

This Subpart prescribes rules governing the issue of and amendment to a provisional type certificate for a product type.

### **21.143. Application for certificate**

An applicant for the issue of a *provisional type certificate* or an amendment to a *provisional type certificate* must submit the application to the Director with a payment of the applicable fee prescribed by regulations made under the Act.

### **21.145. Issue of certificate**

(a) The Director may issue a *provisional type certificate* or an amendment to a *provisional type certificate* and an applicant is entitled to a *provisional type certificate* or an amendment to a *provisional type certificate* if the Director is satisfied that—

- (1) the applicant holds, or has applied for, a type certificate or a supplemental type certificate for the product as appropriate; and

- (2) the applicant meets the applicable requirements of this Subpart; and
- (3) the issue of a *provisional* type certificate or the amendment to a provisional type certificate is not contrary to the interests of aviation safety.

(b) A *provisional* type certificate issued by the Director under this Subpart may be subject to conditions that the Director considers appropriate in each particular case.

#### **21.147. Provisional type certification requirements**

(a) An applicant for the issue of a *provisional* type certificate or an amendment to a provisional type certificate must provide the Director with evidence that—

- (1) there is no feature, characteristic, or condition of the product that makes it unsafe when operated in accordance with the limitations established under paragraph (b)(1); and
- (2) the product has been designed and constructed in accordance with the airworthiness requirements applicable to the issue of the type certificate or supplemental type certificate the applicant holds or has applied for; and
- (3) the product can be operated safely under the appropriate operating limitations established under paragraph (b)(1); and
- (4) for an aircraft type—
  - (i) the aircraft substantially meets the applicable flight characteristic requirements for the type certificate or supplemental type certificate that the applicant holds or has applied for; and
  - (ii) the aircraft has performed every manoeuvre necessary to show compliance with the flight requirements for the issue of the type certificate or supplemental type certificate that the applicant holds or has applied for; and
  - (iii) the aircraft can be operated safely in accordance with any applicable limitations contained in the Civil Aviation Rules.

(b) The applicant must establish—

- (1) the limitations required for the issue of the type certificate or supplemental type certificate that the applicant holds or has applied for, including limitations on weights, speed, flight manoeuvres, loading and operation of controls and equipment; and
- (2) an inspection and maintenance program for the continued airworthiness of the product.

(c) An applicant for the issue of a *provisional* type certificate for an aircraft type must demonstrate appropriate operating restrictions for the aircraft for each limitation not established under paragraph (b)(1).



(d) An applicant for the issue of a *provisional* type certificate for an aircraft type must show that a prototype aircraft has been flown for at least 50 hours—

- (1) under a *special category – experimental* airworthiness certificate; or
- (2) under the auspices of a Defence Force of Mongolia, EU, Russia or the United States of America.

(e) The Director may, in writing, reduce the number of flight hours required under paragraph (d) if the application is for an amendment to a *provisional* type certificate.

#### **21.149. Duration**

(a) A *provisional* type certificate remains in force until its expiry date, which must not be more than 24 months from the date of issue, unless the certificate is suspended or revoked.

(b) An amendment to a *provisional* type certificate remains in force for the period that the *provisional* type certificate remains in force under paragraph (a).

### **Subpart G — Reserved**

### **Subpart H — Airworthiness Certificates**

#### **21.171. Purpose**

This Subpart prescribes rules governing the airworthiness certification of an aircraft.

#### **21.173. Certificate categories**

The following airworthiness certificates for Mongolian registered aircraft are issued by the Director under the Act in accordance with the applicable requirements of this Subpart:

- (1) *standard category*;
- (2) *restricted category*;
- (3) *special category — experimental*;
- (4) *special category — exhibition*;
- (5) *special category — amateur-built*;
- (6) *special category — primary*;
- (7) *special category — LSA*;
- (8) *special category — limited*;
- (9) *provisional category*.

**21.175. Application for certificate**

An applicant for the grant of an airworthiness certificate for an aircraft must complete the applicable form and submit it to the Director with a payment of the applicable fee prescribed by regulations made under the Act:

- (1) form CAA 24021/05 for *standard or restricted category*;
- (2) form CAA 24021/06 for *special category*;
- (3) form CAA 24021/14 for *provisional category*.

**21.177. Special conditions for issue of certificate**

(a) An airworthiness certificate for an aircraft may be issued in both the *standard category* and the *restricted category* if the Director is satisfied that—

- (1) the applicable certification requirements for each category under this Subpart are met when the aircraft is configured for the category; and
- (2) the aircraft can be converted from one configuration to the other by removing or adding equipment by simple mechanical means.

(b) Only a *restricted category* airworthiness certificate may be issued for an aircraft if the Director is satisfied that—

- (1) the aircraft is internally equipped for dispensing substances on an agricultural aircraft operation; and
- (2) the extent of the internal equipment makes the aircraft in appropriate for use in an air transport operation.

(c) An airworthiness certificate for an aircraft may only be issued in 1 of the *special categories* listed under rule 21.173 if the Director is satisfied that—

- (1) the applicable certification requirements under this Subpart are met in respect of the purposes for which the aircraft is to be operated; and
- (2) the aircraft is in a condition for safe operation.

(d) The Director may specify in an airworthiness certificate the purpose of the certificate and any associated conditions and limitations for the operation of the aircraft.

**21.179. Duration of certificate**

(a) The following airworthiness certificates for an aircraft remains in force if the maintenance on the aircraft is performed in accordance with the applicable requirements of Parts 91 and 43 unless the certificate has expired, or is suspended or revoked:

- (1) *standard category*;
- (2) *restricted category*;
- (3) *special category*.

- (b) A *standard category* airworthiness certificate for an aircraft and a *restricted category* airworthiness certificate for an aircraft expire if—
- (1) the aircraft ceases to be a Mongolian registered aircraft; or
  - (2) the *type certificate* or the *type acceptance certificate*, as applicable, issued for the aircraft type, its engines, and propellers if applicable, ceases to remain in force; or
  - (3) a new airworthiness certificate in the same category is issued for the aircraft.
- (c) A *special category* airworthiness certificate for an aircraft expires if—
- (1) the aircraft ceases to be a Mongolian registered aircraft; or
  - (2) a new airworthiness certificate in the same category is issued for the aircraft.
- (d) The holder of an airworthiness certificate that has expired under paragraphs (b) or (c) must surrender the certificate to the Director.
- (e) The holder of a *standard category* airworthiness certificate or a *restricted category* airworthiness certificate that has expired under paragraph (b)(2) may apply under rule 21.175 for the grant of an appropriate *special category* airworthiness certificate for the aircraft.
- (f) A *provisional category* airworthiness certificate for an aircraft remains in force for the period specified in the certificate—
- (1) if—
    - (i) the maintenance on the aircraft is performed in accordance with the applicable requirements of Parts 91 and 43; and
    - (ii) the aircraft remains a Mongolian registered aircraft; and
    - (iii) the *provisional type certificate* issued under rule 21.145 for the aircraft type remains in force; and
  - (2) unless—
    - (i) the *provisional category* airworthiness certificate is suspended or revoked; or
    - (ii) a *standard category* airworthiness certificate or a *restricted category* airworthiness certificate is issued for the aircraft.

### **21.181. Transfer of certificate**

The holder of a Mongolian certificate of registration for an aircraft must transfer the following applicable airworthiness certificates with the aircraft if, in accordance with rule 47.57, the holder of the certificate of registration ceases to have lawful entitlement to possession of the aircraft:

- (1) *standard category*;
- (2) *restricted category*;
- (3) *special category*;
- (4) *provisional category*.

## Certification Requirements

### 21.191. Standard and restricted category requirements

An applicant for a *standard category*, *restricted category*, or *provisional category* airworthiness certificate for an aircraft must provide the Director with evidence that—

- (1) the aircraft, its engines, and propellers if applicable conform to applicable current—
  - (i) type certificates issued in accordance with Subpart B; or
  - (ii) type acceptance certificates issued in accordance with Subpart B; or
  - (iii) provisional type certificates issued in accordance with Subpart F; and
- (2) every modification and repair to the aircraft conforms to a design change approved in accordance with acceptable technical data under Subpart N for the aircraft type; and
- (3) every airworthiness directive that is applicable to the aircraft has been complied with in accordance with Part 39; and
- (4) the aircraft is issued with —
  - (i) the appropriate flight manual; and
  - (ii) the appropriate logbooks, repair and alteration forms and documents; and
- (5) the aircraft is a Mongolian registered aircraft and, in accordance with Part 47, displays an identification plate and 1 of the following forms of marking:
  - (i) the nationality and registration marks;
  - (ii) Police marks;
  - (iii) an approved identifiable paint scheme and markings; and
- (6) the aircraft, its engines, propellers, and propeller hubs and blades are identified by the means specified in Subpart Q; and
- (7) the aircraft conforms with every applicable additional airworthiness requirement prescribed in Part 26; and

- (8) the aircraft has, within 60 days before the application for the airworthiness certificate, undergone —
  - (i) a 100-hour, or equivalent inspection in accordance with the manufacturer's maintenance schedule; or
  - (ii) a scheduled inspection in accordance with an applicable maintenance program of an air operator certificated in accordance with Part 119 or an air transport operator certificated by an ICAO Contracting State; or
  - (iii) an equivalent inspection acceptable to the Director; and
- (9) the aircraft has been weighed within the last 5 years prior to the application under rule 21.175; and
- (10) the aircraft is in a condition for safe operation; and
- (11) the aircraft and each modification and repair to the aircraft complies with the applicable aircraft noise and engine emission standards specified in Appendix C.

### **21.193. Special category—experimental certification requirements**

An applicant for the grant of a *special category—experimental* airworthiness certificate for an aircraft must provide the Director with—

- (1) statement specifying the purpose for which the aircraft is to be operated, including—
  - (i) details of the proposed operation; and
  - (ii) the estimated duration or number of flights required for the operation; and
  - (iii) details of the areas over which the operation is to be conducted; and
- (2) sufficient data to identify the aircraft, including a three-view drawing of the aircraft, except for an aircraft converted from a previously certificated type without appreciable change in the external configuration; and
- (3) any other information that the Director may request relating to the application; and
- (4) flight manuals, maintenance manuals and any document relating to the operation of the aircraft as the Director may require; and
- (5) evidence that the aircraft conforms with any design change necessary for the safe operation of the aircraft that the Director may require; and
- (6) evidence that the aircraft is a Mongolian registered aircraft and, in accordance with Part 47, displays an identification plate and 1 of the following forms of marking—
  - (i) the nationality and registration marks; or
  - (ii) an approved identifiable paint scheme and markings; and

- (7) evidence that the aircraft is identified by the means specified in Subpart Q; and
- (8) evidence that a maintenance program for the aircraft has been approved in accordance with rule 91.607; and
- (9) evidence that the placard requirements prescribed in rule 21.205 are met.

#### **21.195. Special category—exhibition certification requirements**

(a) An applicant for the grant of a *special category—exhibition* airworthiness certificate for an aircraft, must provide the Director with—

- (1) a statement specifying the purpose for which the aircraft is to be operated; and
- (2) sufficient data to identify the aircraft make and model; and
- (3) any other information that the Director may request relating to the application; and
- (4) flight manuals, maintenance manuals and any documents relating to the operation of the aircraft as the Director may require; and
- (5) evidence that the aircraft conforms with any design change necessary for the safe operation of the aircraft that the Director may require; and
- (6) evidence that the aircraft is a Mongolian registered aircraft and, in accordance with Part 47, displays an identification plate and 1 of the following forms of marking—
  - (i) the nationality and registration marks; or
  - (ii) an approved identifiable paint scheme and markings; and
- (7) evidence that the holder of the certificate of registration for the aircraft holds a valid operator statement for the aircraft in accordance with rule 47.55(c); and
- (8) evidence that the aircraft is identified by the means specified in Subpart Q; and
- (9) evidence that a flight evaluation has been completed and that the aircraft—
  - (i) is controllable throughout its range of speeds and throughout all the normal operating manoeuvres; and
  - (ii) does not have any hazardous operating characteristics or design features; and
- (10) evidence that a maintenance program for the aircraft has been approved in accordance with rule 91.607; and
- (11) evidence that every applicable airworthiness directive has been complied with in accordance with Part 39; and

- (12) data from a flight evaluation concerning—
  - (i) any operational limitations for the aircraft; and
  - (ii) any airworthiness limitations for the aircraft; and
- (13) evidence that the placard requirements prescribed in rule 21.205 are met.
- (b) A flight evaluation required by paragraph (a)(9) must be conducted—
  - (1) in accordance with the limitations and conditions stated on a *special category—experimental* airworthiness certificate issued for the aircraft; and
  - (2) by a test pilot approved for the purpose under rule 19.405.

### **21.197. Special category—amateur-built certification requirements**

- (a) An applicant for the grant of a special *category—amateur—built* airworthiness certificate for an aircraft must provide the Director with—
  - (1) a statement specifying the purpose for which the aircraft is to be operated; and
  - (2) sufficient data to identify the aircraft make and model; and
  - (3) any other information that the Director may request relating to the application; and
  - (4) flight manuals, maintenance manuals and any documents relating to the operation of the aircraft as the Director may require; and
  - (5) evidence that—
    - (i) the amateur constructor has built the major portion of the aircraft for their own education or recreation; or
    - (ii) the aircraft is a light sport aircraft designated by the manufacturer as an experimental – light sport aircraft; and
  - (6) evidence that the aircraft conforms with any design change necessary for the safe operation of the aircraft that the Director may require; and
  - (7) evidence that the aircraft is a Mongolian registered aircraft and, in accordance with Part 47, displays an identification plate and 1 of the following forms of marking—
    - (i) the nationality and registration marks; or
    - (ii) an approved identifiable paint scheme and markings; and
  - (8) evidence that the aircraft is identified by the means specified in Subpart Q; and

- (9) evidence that a flight evaluation has been completed and that the aircraft—
    - (i) is controllable throughout its range of speeds and throughout all the normal operating manoeuvres; and
    - (ii) does not have any hazardous operating characteristics or design features; and
  - (10) evidence that a maintenance program for the aircraft has been approved in accordance with rule 91.607; and
  - (11) evidence that every applicable airworthiness directive has been complied with in accordance with Part 39; and
  - (12) data from a flight evaluation concerning—
    - (i) any operational limitations for the aircraft; and
    - (ii) any airworthiness limitations for the aircraft; and
  - (13) evidence that the placard requirements prescribed in rule 21.205 are met.
- (b) A flight evaluation required by paragraph (a)(12) must be conducted—
- (1) in accordance with the limitations and conditions stated on a *special category—experimental* airworthiness certificate issued for the aircraft; and
  - (2) by a test pilot approved for the purpose under rule 19.405.

#### **21.199. Special category—primary certification requirements**

An applicant for the grant of a *special category—primary* airworthiness certificate for an aircraft must provide the Director with—

- (1) a statement specifying the purpose for which the aircraft is to be operated; and
- (2) sufficient data to identify the aircraft make and model; and
- (3) any other information that the Director may request relating to the application; and
- (4) flight manuals, maintenance and repair manuals, parts manual, and any other documents relating to the operation of the aircraft as the Director may require; and
- (5) evidence that every modification and repair to the aircraft conforms to a design change approved in accordance with Subpart C; and
- (6) evidence that every applicable airworthiness directive has been complied with in accordance with Part 39; and
- (7) evidence that the aircraft is a Mongolian registered aircraft and, in accordance with Part 47, displays an identification plate and 1 of the following forms of marking—
  - (i) the nationality and registration marks; or



- (ii) an approved identifiable paint scheme and markings; and
- (8) evidence that the aircraft, its engines, propellers, and propeller hubs and blades are identified by the means specified in Subpart Q; and
- (9) evidence that a maintenance program for the aircraft has been approved under rule 91.607; and
- (10) evidence that continuing airworthiness data relating to the aircraft and installed components is available from sources acceptable to the Director; and
- (11) evidence that the aircraft has, within 60 days prior to the application for the airworthiness certificate, undergone a 100- hour, or equivalent inspection in accordance with the manufacturer's maintenance schedule, or an equivalent inspection acceptable to the Director; and
- (12) for an imported aircraft—
  - (i) evidence that the aircraft had been certificated as a primary category aircraft in accordance with FAR 21.24, or an equivalent certification acceptable to the Director, by the relevant authority for airworthiness in the country of export; and
  - (ii) evidence that the aircraft conforms to the requirements and airworthiness design standards for a primary category aircraft prescribed in FAR 21.24, or the requirements and airworthiness design standards for the equivalent certification accepted by the Director under paragraph (i); and
- (13) for a Mongolian manufactured aircraft—
  - (i) evidence that the aircraft was manufactured under the authority of an aircraft manufacturing organisation certificate issued in accordance with Part 148; and
  - (ii) evidence that the aircraft conforms with the requirements and airworthiness design standards for a primary category aircraft prescribed in FAR 21.24, or conforms with an equivalent airworthiness design standard acceptable to the Director; and
- (14) evidence that the placard requirements prescribed in rule 21.205 are met.

#### **21.201. Special category—LSA certification requirements**

- (a) An applicant for the grant of a *special category—LSA* airworthiness certificate for an aircraft must provide the Director with—
- (1) a statement specifying the purpose for which the aircraft is to be operated; and
  - (2) sufficient data to identify the aircraft make and model; and
  - (3) any other information that the Director may request relating to the application; and

- 
- (4) flight manuals, the manufacturer's maintenance manual and inspection procedures, and any other documents relating to the operation of the aircraft as the Director may require; and
  - (5) evidence that the aircraft conforms with any design change necessary for the safe operation of the aircraft that the Director may require; and
  - (6) evidence that the aircraft is a Mongolian registered aircraft and, in accordance with Part 47, displays an identification plate and 1 of the following forms of marking—
    - (i) the nationality and registration marks; or
    - (ii) an approved identifiable paint scheme and markings; and
  - (7) evidence that the aircraft is identified by the means specified in Subpart Q; and
  - (8) evidence that a flight evaluation has been completed and that the aircraft—
    - (i) is controllable throughout its range of speeds and throughout all the normal operating manoeuvres; and
    - (ii) does not have any hazardous operating characteristics or design features; and
  - (9) evidence that a maintenance program for the aircraft has been approved in accordance with rule 91.607; and
  - (10) evidence that every applicable airworthiness directive has been complied with in accordance with Part 39; and
  - (11) for an imported aircraft—
    - (i) evidence that the aircraft has been issued with, or is eligible for the issue of, a special airworthiness certificate in the light sport category in accordance with the requirements specified in FAR 21.190 or an equivalent standard acceptable to the Director; and
    - (ii) the aircraft manufacturer's signed statement of compliance for the aircraft, or a certified copy of the manufacturer's signed statement, that meets the requirements of paragraph (b); and
  - (12) for a Mongolian manufactured aircraft—
    - (i) evidence that the aircraft was manufactured under the authority of an aircraft manufacturing organisation certificate issued in accordance with Part 148; and
    - (ii) the aircraft manufacturer's signed statement of compliance for the aircraft, or a certified copy of the manufacturer's signed statement, that meets the requirements of paragraph (b); and
  - (13) evidence that the placard requirements prescribed in rule 21.205 are met.

(b) A manufacturer's statement of compliance required under paragraphs (a)(11) and (12) must include at least the following information:

- (1) details specifying the aircraft's make and model, serial number and date of manufacture;
- (2) technical details to show that the aircraft can be defined as a light sport aircraft;
- (3) the identification of the consensus standard that is applicable to the design and manufacture of the aircraft;
- (4) a statement certifying that—
  - (i) the manufacturer has a quality assurance system that meets the requirements of the consensus standard identified under paragraph (b)(3); and
  - (ii) the manufacturer's design data for the aircraft conforms with the consensus standard identified under paragraph (b)(3); and
  - (iii) based on the quality assurance system required under paragraph (b)(4)(i), the aircraft conforms with the manufacturer's design data for the aircraft and with the consensus standard identified under paragraph (b)(3); and
  - (iv) in accordance with a production acceptance test procedure established by the manufacturer in accordance with the consensus standard identified under paragraph (b)(3)—
    - (A) the manufacturer has ground-tested and flight-tested the aircraft; and
    - (B) the manufacturer is satisfied that the aircraft's performance during ground and flight testing meets the design requirements; and
    - (C) the aircraft is in a condition for safe operation; and
  - (v) the manufacturer has a procedure to monitor the continuing airworthiness of the aircraft and to issue safety directives or requirements that comply with the consensus standard identified under paragraph (b)(3) to correct any unsafe condition.

(c) A light sport aircraft that is built from a kitset and a light sport aircraft that is designated by the manufacturer as an experimental - light sport aircraft is not eligible for the issue of a *special category*—LSA airworthiness certificate.

**21.203. Special category—limited certification requirements**

(a) An applicant for the grant of a *special category—limited* airworthiness certificate for an aircraft must provide the Director with—

- (1) a statement specifying the purpose for which the aircraft is to be operated; and
- (2) sufficient data to identify the aircraft make and model; and
- (3) evidence that the aircraft type was produced in a series and factory built in a controlled design environment; and
- (4) evidence that the aircraft type has been previously accepted for civil or military operations; and
- (5) evidence that the aircraft—
  - (i) is essentially unmodified from the original type design and operational configuration; or
  - (ii) has been extensively restored without significant variation to the original type design and operational configuration; or
  - (iii) essentially conforms with the original design standards and operational configuration in spite of having significant changes made which are subject to an appropriate design approval or are acceptable to the Director; and
- (6) evidence that every applicable airworthiness directive has been complied with in accordance with Part 39; and
- (7) evidence that a maintenance program for the aircraft has been approved in accordance with rule 91.607; and
- (8) flight manual limitation data including the following:
  - (i) speed parameters:
  - (ii) weight and loading information:
  - (iii) engine operating parameters:
  - (iv) fuel and oil specifications:
  - (v) fuel consumption data:
  - (vi) total usable fuel quantities:
  - (vii) eligibility for flight in known icing conditions:
  - (viii) critical and prohibited manoeuvres:
  - (ix) emergency procedures:
  - (x) maximum number of occupants:
  - (xi) minimum flight crew members if more than one flight crew member is required:

- (xii) recommended take-off and landing distances:
  - (xiii) if applicable, single engine climb performance; and
- (9) evidence that the aircraft is a Mongolian registered aircraft and, in accordance with Part 47, displays an identification plate and 1 of the following forms of marking—
- (i) the nationality and registration marks; or
  - (ii) an approved identifiable paint scheme and markings; and
- (10) evidence that the holder of the certificate of registration for the aircraft holds a valid operator statement for the aircraft in accordance with rule 47.55(c); and
- (11) evidence that the aircraft has undergone an inspection, equivalent to an annual or 100-hour inspection in accordance with the manufacturer's or a military maintenance schedule, or an equivalent inspection acceptable to the Director, within the 60 day period before the application for the airworthiness certificate was submitted to the Director; and
- (12) evidence that the aircraft has completed—
- (i) a flight evaluation to establish that the aircraft is controllable throughout its range of speeds and throughout all the normal operating manoeuvres if documented evidence of the aircraft's safety, integrity and reliability record is not available; or
  - (ii) a flight evaluation to establish known operating limitations or confirm changes to those limitations as a result of modification; or
  - (iii) a combination of paragraphs (i) and (ii); and
- (13) evidence that the aircraft is in a condition for safe operation; and
- (14) evidence that the placard requirements prescribed in rule 21.205 are met.
- (b) The flight manual limitation data required by paragraph (a)(8) must originate from—
- (1) the manufacturer's data or military operating manuals acceptable to the Director; or
  - (2) data accepted by a foreign civil airworthiness authority; or
  - (3) data generated from a flight evaluation conducted under a *special category—experimental* airworthiness certificate issued for the aircraft; or
  - (4) a combination of paragraphs (1), (2), and (3).
- (c) A flight evaluation required by paragraphs (a)(12), and (b)(3) must be conducted—

- (1) in accordance with the limitations and conditions stated on a *special category—experimental* airworthiness certificate issued for the aircraft; and
- (2) by a test pilot approved for the purpose under rule 19.405.

### **21.205. Placard for special category aircraft**

An aircraft that is issued with a *special category* airworthiness certificate must have a legible placard with the following wording, or other similar wording acceptable to the Director, installed in the aircraft and in clear view of each passenger—

#### ***“Warning***

***This is a Special Category — (Subcategory) aircraft and does not meet the Mongolian Civil Aviation airworthiness standards for a standard category aircraft to carry passengers on an air transport operation.***

***Passengers fly in this aircraft at their own risk.”***

## **Subpart I — Special Flight Permits**

### **21.251. Applicability**

This Subpart prescribes rules governing the issue of a special flight permit for an aircraft.

### **21.253. Special flight permit categories**

The following special flight permits for an aircraft are granted by the Director under the Act and in accordance with the applicable requirements of this Subpart:

- (1) *special flight permit*:
- (2) *special flight permit—continuing authorisation*.

### **21.255. Application for special flight permit**

An applicant for the grant of a special flight permit must complete 1 of the following forms as appropriate and submit it to the Director with a payment of the applicable fee prescribed by regulations made under the Act—

- (1) form CAA 24021/07 for a *special flight permit*:
- (2) form CAA 24021/15 for a *special flight permit—continuing authorisation*.

**21.257. Conditions for issue of special flight permit**

(a) The Director may issue a *special flight permit* for an aircraft if the Director is satisfied that—

- (1) the applicable certification requirements under this Subpart are met; and
- (2) in spite of any failure of the aircraft to meet applicable airworthiness and maintenance requirements prescribed in this Part or any other civil aviation rule, the level of safety is adequate for any of the following purposes:
  - (i) flying the aircraft to a base where maintenance is to be performed, or to a point of storage;
  - (ii) flight testing of a new aircraft by the holder of a manufacturing organisation certificate issued in accordance with Part 148;
  - (iii) evacuating the aircraft from an area of impending danger.

(b) The Director may issue a *special flight permit—continuing authorisation* for an aircraft type if the Director is satisfied that the applicable certification requirements under this Subpart are met and the level of safety is adequate for the purposes for which an aircraft is to be operated.

(c) The Director may specify in a special flight permit the purpose of the permit and any associated conditions and limitations for the operation of the aircraft.

**21.259. Duration of special flight permit**

(a) A *special flight permit* for an aircraft remains in force for the period specified in the permit if the aircraft remains a Mongolian registered aircraft, unless the permit is suspended or revoked by the Director.

(b) A *special flight permit—continuing authorisation* for an aircraft type remains in force for the period specified in the permit unless the permit is suspended or revoked by the Director.

**Special Flight Permit Requirements****21.261. Application for special flight permit**

An applicant for the grant of a *special flight permit* for an aircraft must provide the Director with—

- (1) details of—
  - (i) the purpose of the flight; and
  - (ii) the proposed itinerary; and
  - (iii) the crew members required to operate the aircraft and its equipment; and

- (iv) any non-compliance with any applicable airworthiness or maintenance requirement; and
  - (v) any restriction that may be necessary for the safe operation of the aircraft; and
- (2) any other information the Director may require; and
  - (3) evidence that the aircraft is a Mongolian registered aircraft and, in accordance with Part 47, displays an identification plate and an approved form of aircraft marking; and
  - (4) evidence that the aircraft is identified by a means specified in Subpart Q.

### **21.263. Application for special flight permit—continuing authorisation**

An applicant for the grant of a *special flight permit—continuing authorisation* for an aircraft type must provide the Director with—

- (1) a statement to confirm that the operation of any aircraft under the special flight permit is for the purpose of flight testing under a manufacturing organisation certificate issued in accordance with Part 148; and
- (2) details of the aircraft type; and
- (3) details of the manufacturing organisation's procedures for ensuring that each flight test is carried out safely; and
- (4) any other information the Director may require.

## **Subpart J — Reserved**

## **Subpart K — Materials, Parts, Processes, and Appliances**

### **21.301. Purpose**

This Subpart prescribes rules governing the approval or acceptance of a material, part, process, and appliance.

### **21.303. Replacement and modification materials, parts, and appliances**

A replacement or modification material, part, or appliance to be installed into a type certificated product must—

- (1) be authorised by the holder of the type certificate for the product as complying with the type design; or
- (2) be manufactured by a person performing maintenance on the product and the replacement or modification material, part, or appliance must be —



- (i) certified by the person to conform to an approved type design including any embodied design change; and
  - (ii) identified in accordance with Subpart Q, or
- (3) be issued with a *CAA Form One – authorised release certificate* issued by the holder of a maintenance organisation certificate issued in accordance with Part 145 who established the conformity and traceability of, and performed maintenance on, the part or appliance; or
- (4) be issued with a *CAA Form One – authorised release certificate* issued by the holder of a manufacturing organisation certificate issued in accordance with Part 148 who manufactured the part or appliance in accordance with a—
- (i) Mongolian TSO authorisation; or
  - (ii) Mongolian PMA authorisation; or
  - (iii) supplemental type certificate issued under Subpart E; or
- (5) be issued with a release note, by the holder of a supply organisation certificate issued in accordance with Part 145, who established the conformity of the material, part, or appliance; or
- (6) be a standard part; or
- (7) be an imported part accepted for the purpose by the Director.

### **21.305. Approval or acceptance of processes**

If a process is required to meet the airworthiness requirements for the issue of a type certificate or of any CAR, then it shall be—

- (1) included in the exposition of the holder of a maintenance organisation certificate issued under Part 145; or
- (2) included in the exposition of the holder of a manufacturing organisation certificate issued under Part 148; or
- (3) otherwise acceptable to the Director.

## **Subpart L — Export Airworthiness Certificates**

### **21.321. Purpose**

This Subpart prescribes rules governing the issue of—

- (1) an export airworthiness certificate for the export of a product:
- (2) *CAA Form One – authorised release certificates* for the export of a part and an appliance.

**21.323. Reserved****21.325. Reserved****21.327. Export airworthiness certificate exceptions**

If the applicant for an export airworthiness certificate provides a written statement by the State of the importer, in accordance with 21.333(b), the Director may issue the export airworthiness certificate with listed exceptions of—

- (1) the requirements of this Subpart that have not been met; and
- (2) any differences in configuration between the exported product and the relative type approved or type accepted product.

**21.329. Eligibility**

Any exporter or exporter's authorised representative may apply for the issue of an export airworthiness certificate for a product.

**21.331. Application for export airworthiness certificate**

(a) An applicant for the issue of an *export* airworthiness certificate must complete form CAA 24021/10, and submit it to the Director with a payment of the applicable fee prescribed by regulations made under the Act and provide the Director with—

- (1) evidence that—
  - (i) the product conforms to a type design acceptable to the State of the importer; and
  - (ii) any special certification condition of the State of the importer has been met; and
  - (iii) the State of the importer accepts any exception to be listed in the certificate; and
  - (iv) the product has been identified in accordance with Subpart Q; and
  - (v) the applicable airworthiness directives have been complied with; and
- (2) any log book, modification and repair form, and historical record that the Director may require for other than a new product; and
- (3) a description of any method used, including the duration of the effectiveness of the method, for the preservation and packaging of a product, to protect it against corrosion and damage while in transit or storage; and
- (4) the date when ownership passed, or is expected to pass, to a purchaser in the State of the importer; and

- (5) the date on which any document that is not available at the date of application is expected to become available; and
  - (6) supporting documentation for any variance to this Subpart; and
  - (7) further particulars relating to the product and the applicant, if required by the Director as indicated in the form.
- (b) An applicant for the issue of an export airworthiness certificate for an aircraft must, in addition to paragraph (a), provide the Director with—
- (1) evidence that—
    - (i) for a new aircraft, it has been manufactured under the authority of a manufacturing organisation certificate issued in accordance with Part 148; and
    - (ii) for other than a new aircraft, it possesses or qualifies for an airworthiness certificate under Subpart H; and
    - (iii) the aircraft is issued with the appropriate flight manual and, for a new aircraft, maintenance manual; and
    - (iv) a weight and balance report has been completed, with a loading schedule if applicable; and
    - (v) the aircraft has, within 60 days before the application for the export airworthiness certificate, undergone a 100- hour, or equivalent, inspection in accordance with a manufacturer's maintenance schedule, or an equivalent inspection acceptable to the Director; and
    - (vi) any installations incorporated for the purpose of export delivery comply with the applicable airworthiness requirements or have been approved by the issue of a *special category – special flight permit* airworthiness certificate under Subpart H; and
  - (2) confirmation that any installation described in paragraph (b)(1)(vi) is to be removed and the aircraft restored to the approved type configuration upon completion of the delivery flight.
- (c) The applicant must make the product that is the subject of the export airworthiness certificate and associated data available for inspection if required by the Director.

### **21.333. Issue of export airworthiness certificate**

- (a) The Director may issue an export airworthiness certificate for a product and an applicant is entitled to an export airworthiness certificate if the Director is satisfied that—
- (1) the applicant meets the applicable requirements of this Subpart; and
  - (2) the issue of the certificate is not contrary to the interests of aviation safety; and

- (3) any airworthiness requirement that is not complied with is compensated for by a factor that provides an acceptable level of safety.
- (b) Notwithstanding paragraph (a)(1), the Director may issue an export airworthiness certificate for a product that does not meet every airworthiness requirement of rule 21.331 if the applicant provides written evidence that the non-compliance with any particular requirement is acceptable to the State of the importer.
- (c) An export airworthiness certificate issued by the Director under this Subpart—
- (1) may be subject to conditions as the Director considers appropriate in each particular case; and
  - (2) does not authorise the installation or use of a product.

### **21.335. Validity of certificate**

- (a) An export airworthiness certificate issued under this Subpart shall remain valid, providing there is no subsequent design change to the product, until the completion of delivery to the importer's State.
- (b) The holder of an export airworthiness certificate invalidated because of a design change shall forthwith surrender the certificate to the Director.

### **21.337. Transfer of certificate**

An export airworthiness certificate is transferred with the product.

### **21.339. Use of CAA Form One – authorised release certificate for export**

- (a) The CAA *Form One – authorised release* certificate must only be used for the export of a part or appliance if—
- (1) the part or appliance—
    - (i) is new, has been newly overhauled, or was last installed in an aircraft which possesses a valid standard or restricted category airworthiness certificate and the part or appliance is fit for release to service; and
    - (ii) conforms to approved design data; and
    - (iii) is in a condition for safe operation; and
    - (iv) meets any special conditions for import required by the State of the importer; and
  - (2) the CAA *Form One – authorised release* certificate has been issued in accordance with the procedures of—
    - (i) an aircraft maintenance organisation certificated in accordance with Part 145; or

- (ii) an aircraft manufacturing organisation certificated in accordance with Part 148.

(b) A part or appliance does not need to meet every requirement under paragraph (a) if the State of the importer indicates in writing that the part or appliance is acceptable to the State.

#### **21.341. Responsibilities of an exporter**

(a) When title to an aircraft passes or has passed to a foreign purchaser, the exporter who was granted an export airworthiness certificate shall—

- (1) where applicable, request the cancellation of the Mongolian registration and airworthiness certificates, giving the date of transfer of title and the name and address of the foreign owner; and
- (2) return the registration and airworthiness certificates to the Director; and
- (3) submit a statement certifying that the Mongolian nationality and registration marks have been removed from the aircraft.

(b) Unless otherwise agreed with the State of the importer, the exporter who was granted an export airworthiness certificate shall—

- (1) forward to the appropriate authority of the State of the importer—
  - (i) all documents and information necessary for the proper operation of the product and any other material as is stipulated in the special requirements of the State of the importer; and
  - (ii) the applicable manufacturer's assembly instructions for un-assembled aircraft and an approved flight test check list; and
- (2) preserve and package products to protect them against corrosion and damage whilst in transit or storage; and
- (3) upon completion of an export delivery of an aircraft, remove, or have removed, any temporary installation incorporated for the purpose of delivery and restore the aircraft to the approved type configuration.

## **Subpart M — Repairs**

#### **21.431. Applicability**

This Subpart prescribes the means for the approval of repair designs.

**21.433. Approval of designs for repairs**

A repair, the design of which has not been approved under the type certificate or type acceptance certificate, shall be treated as a design change to be approved in accordance with Subparts C, D, or E.

**Subpart N — Technical Data and Airworthiness Specifications****21.501. Purpose**

This Subpart prescribes rules governing—

- (1) the approval or acceptance of technical data:
- (2) the approval or acceptance of specifications for a material, part, process, or appliance:
- (3) the approval of a deviation from specifications for a material, part, process, or appliance.

**21.503. Acceptable technical data**

- (a) Technical data shall only be used if it is approved, or is acceptable to the Director.
- (b) Acceptable technical data are listed in Appendix D to this Part.

**21.505 Form CAA 337 – approval of technical data**

(a) Except as provided in paragraphs (b) and (c), an applicant for the approval of technical data must complete form CAA 337, and submit it to the Director with a payment of the applicable fee prescribed by regulations made under the Act and provide the Director with—

- (1) the name and address for service in Mongolia of the applicant; and
- (2) any documentation necessary to define the data; and
- (3) a description of any design change including—
  - (i) sufficient data to identify the change; and
  - (ii) the identification of every part of a product, component, or appliance affected by the change; and
- (4) for a product, component, or appliance to be changed in accordance with the data,—
  - (i) details of any investigation, test or analysis that may be necessary to show compliance with the applicable airworthiness requirements; and

- (ii) adequate maintenance and operating data to ensure the product, component, or appliance can be properly maintained and operated; and
- (5) any further particulars relating to the applicant, the technical data, and any design change, if required by the Director as indicated in the form; and
- (6) one of the following:
  - (i) a statement of compliance provided by a design organisation certificated in accordance with Part 146 stating that the technical data meets the airworthiness requirements of rule 21.31, and the applicable additional airworthiness requirements prescribed in Part 26, and the applicable aircraft noise and engine emission requirements prescribed in rule 21.32, and that the associated design change is fit for embodiment:
  - (ii) a statement equivalent to that required by paragraph (a)(6)(i) issued in accordance with the requirements of an appropriate foreign authority:
  - (iii) a written request that a statement of compliance be provided during the technical data approval process.
- (b) Notwithstanding paragraph (a)(6)(i), a statement of compliance may state that the technical data—
  - (1) meets the airworthiness, aircraft noise and engine emission requirements incorporated by reference in the type certificate; or
  - (2) if special conditions are applied, meets a level of safety equivalent to that required under paragraph (a)(6)(i).
- (c) An aircraft design organisation certificated in accordance with Part 146 may use a document other than form CAA 337 to record the information required under paragraph (a).
- (d) The Director may approve technical data and an applicant is entitled to the approval of technical data if the Director is satisfied that—
  - (1) the applicant meets the applicable requirements under paragraph (a); and
  - (2) the approval of the data is not contrary to the interests of aviation safety; and
  - (3) any airworthiness requirement that is not complied with is compensated for by a factor that provides an acceptable level of safety; and
  - (4) there is no feature or characteristic of a product, component, or appliance that makes it unsafe for its intended use when the product, component, or appliance is—
    - (i) changed in accordance with the data; and

- (ii) operated in accordance with the correctly amended flight manual or other specified limitations.
- (e) Notwithstanding paragraph (d)(1), the Director may approve technical data without the applicant meeting every requirement of paragraph (a) if—
- (1) the application is limited to an individual product, component, or appliance; and
  - (2) the technical data for an aircraft that holds a special category airworthiness certificate meets the standards that applied for the issue of the airworthiness certificate; and
  - (3) the data provided by the applicant is sufficient to identify the associated design change.

#### **21.507. Approval of specifications**

The specification for a material, part, process, or appliance shall be approved by—

- (1) the issue of an Mongolian TSO if sufficient information is available to show that the specification defines an acceptable minimum performance standard; or
- (2) the application of that specification in an approved design or design change but only for that design or design change; or
- (3) another method acceptable to the Director.

#### **21.509. Acceptance of specifications**

A specification for a material, part, process, or appliance may be acceptable to the Director if—

- (1) it is applied by, or accepted for use in, an airworthiness design standard listed in Appendix C and is used only for that airworthiness design standard; or
- (2) it has been approved or accepted by a foreign aviation authority; or
- (3) it is a specification for a standard part and it is—
  - (i) an established industry specification; or
  - (ii) a Mongolian national specification; or
  - (iii) a foreign national specification.

#### **21.511. Application for deviation from specification**

(a) Each applicant for a deviation to the performance standard of a Mongolian TSO or accepted specification shall complete form CAA 24021/11, which shall require—



- (1) the name and address for service in Mongolia of the applicant; and
- (2) the identification of the product, component, or appliance to which the deviation is to apply; and
- (3) any documentation necessary to support the deviation and its suitability for application to the product, component, or appliance; and
- (4) evidence that the standard from which a deviation is requested is compensated for by factors or design features providing—
  - (i) an equivalent minimum performance standard; and
  - (ii) a level of safety acceptable to the Director; and
- (5) such further particulars relating to the applicant as may be required by the Director as indicated on the form—

and submit it to the Director with a payment of the applicable fee prescribed by regulations made under the Act.

- (b) An applicant is entitled to an approval of a deviation from specification if—
  - (1) the applicant meets the applicable requirements of this Rule in a manner acceptable to the Director; and
  - (2) the approval of the deviation is not contrary to the interests of aviation safety; and
  - (3) any airworthiness requirements not complied with are compensated for by factors that provide a level of safety acceptable to the Director.

## Subpart O — Mongolian Technical Standard Order Authorisations

### 21.601. Applicability

This Subpart prescribes rules governing—

- (1) the approval of designs for articles by the issue of a Mongolian TSO authorisation; and
- (2) the responsibilities of the holder of a Mongolian TSO authorisation.

### 21.603. Definitions

In this Part:

**Article** means any material, part, process, or appliance.

### 21.605. Requirement for Mongolian TSO authorisation

- (a) A Mongolian TSO authorisation is—
  - (1) a design approval issued to the manufacturer of an article that has been found to meet a Mongolian TSO; and

(2) an authorisation to mark an article with the letters Mongolian TSO.

(b) No person may identify an article with a Mongolian TSO marking unless that person holds a Mongolian TSO authorisation in respect of that article and the article meets the Mongolian TSO performance standards.

#### **21.607. Eligibility**

(a) The applicant for the grant of a Mongolian TSO authorisation shall—

(1) be the holder of, or the applicant for, a manufacturing organisation certificate issued under Part 148; and

(2) have made all necessary preparations to, or be able to, produce the article.

(b) In addition to paragraph (a), if the Director determines that the specification includes significant airworthiness requirements, the applicant for the grant of a Mongolian TSO authorisation shall—

(1) be the holder of, or the applicant for, a design organisation certificate issued under Part 146 with a rating appropriate to the article to be produced; or

(2) arrange for the holder of a design organisation certificate with a rating appropriate to the article to be produced to show compliance with the specification; or

(3) have an equivalent procedure acceptable to the Director.

#### **21.609. Application for authorisation**

Each applicant for the grant of a Mongolian TSO authorisation shall complete form CAA 24021/12, which shall require—

(1) the name and address for service in Mongolia of the applicant; and

(2) evidence that the applicant can—

(i) comply with the design and performance requirements of the Mongolian TSO; and

(ii) reliably reproduce each article in accordance with the technical conditions of the applicable specification; and

(3) one copy of the technical data required by the applicable Mongolian TSO; and

(4) a statement of compliance by the applicant, or for an article that the Director determines includes significant airworthiness requirements, by the holder of a design organisation certificate issued under Part 146, stating that each article to be produced will comply with the technical conditions of the applicable specification; and

(5) such further particulars relating to the applicant as may be required by the Director as indicated on the form—

and submit it to the Director with a payment of the applicable fee prescribed by regulations made under the Act.

#### **21.611. Issue of authorisation**

An applicant is entitled to a Mongolian TSO authorisation if—

- (1) the applicant meets the applicable requirements of this Subpart in a manner acceptable to the Director; and
- (2) the granting of the certificate is not contrary to the interests of aviation safety; and
- (3) the applicant has obtained a manufacturing organisation certificate issued under Part 148.

#### **21.613. Privileges of authorisation**

(a) The holder of a Mongolian TSO authorisation for an article may—

- (1) manufacture the article subject to compliance with 21.621 and 21.623(a) and (b); and
- (2) identify the article with the Mongolian TSO marking.

#### **21.615. Validity of authorisation**

The Mongolian TSO authorisation may be suspended or revoked if the Director determines that—

- (1) the conditions required under this Subpart for the grant of an authorisation are no longer being observed; or
- (2) the specification has been—
  - (i) cancelled; or
  - (ii) revised to such an extent that the existing authorisation is no longer appropriate; or
- (3) the manufacturing organisation certificate issued under Part 148 is suspended or revoked; or
- (4) the responsibilities of the holder specified in 21.621 are no longer being discharged; or
- (5) the article has been shown to give rise to unacceptable hazards in service.

#### **21.617. Duration of authorisation**

(a) A Mongolian TSO authorisation issued under this Subpart shall remain in force until it is suspended or revoked.

(b) The holder of a Mongolian TSO authorisation that is revoked shall forthwith surrender the certificate to the Director.

(c) The holder of a Mongolian TSO authorisation that is suspended shall forthwith produce the certificate to the Director for appropriate endorsement.

#### **21.619. Transfer of authorisation**

A Mongolian TSO authorisation issued under this Subpart is not transferable.

#### **21.621. Responsibilities of authorisation holder**

Each manufacturer of an article who is the holder of a Mongolian TSO authorisation for that article shall—

- (1) manufacture each part in accordance with the manufacturing organisation certificate issued under Part 148; and
- (2) ensure that each completed article conforms to its design data and is safe for installation; and
- (3) prepare, maintain, and update master copies of all manuals required for the maintenance and operation of the product; and
- (4) make available to the users of the article those maintenance, overhaul, and repair manuals necessary for the maintenance and operation of the article; and
- (5) permanently and legibly mark each article with the following information—
  - (i) the information prescribed in 21.813; and
  - (ii) the name, type, or part number of the article; and
  - (iii) the applicable Mongolian TSO number; and
- (6) upon the Director's request, make the maintenance, overhaul, and repair manuals and any other records required by 21.625 available to the Director.

#### **21.623. Design changes**

(a) No person shall make design changes to an article under this Subpart unless—

- (1) they are the holder of the appropriate Mongolian TSO authorisation; or
- (2) they apply for a separate Mongolian TSO authorisation.

(b) The holder of a Mongolian TSO authorisation may make design changes to an article it manufactures, other than significant design changes, without further authorisation by the Director provided the holder—

- (1) retains the basic model number of the article; and
- (2) identifies any design change by a model number suffix or part number change; and

- (3) forwards to the Director any revised data that is necessary to identify the design change and show continued compliance with the Mongolian TSO.
- (c) The holder of a Mongolian TSO authorisation intending to make a significant design change to an article it manufactures shall—
- (1) assign a new type or model designation to the article; and
  - (2) apply for a new Mongolian TSO authorisation.

#### **21.625. Records**

- (a) Each holder of a Mongolian TSO authorisation shall establish procedures to—
- (1) identify, collect, index, store, maintain, and dispose of—
    - (i) records of each model of each article manufactured by the holder under a Mongolian TSO authorisation; and
    - (ii) all relevant design information, drawings, specifications, tests, analyses, and reports that show the articles comply with the applicable Mongolian TSO and this Part; and
  - (2) ensure that—
    - (i) all records are legible and of a permanent nature; and
    - (ii) except as provided in paragraph (b), the records required by paragraph (1) are retained for a period of 2 years from the date the last example of the article is permanently withdrawn from service.
- (b) The Director may permit records to be retained for a lesser period than that required by paragraph (a)(2)(ii).

### **Subpart P — Mongolian Parts Manufacturing Approval Authorisations**

#### **21.701. Applicability**

This Subpart prescribes rules governing—

- (1) the approval of designs for replacement or modification parts by the issue of a Mongolian PMA authorisation; and
- (2) the responsibilities of the holder of a Mongolian PMA authorisation.

#### **21.703. Requirement for Mongolian PMA authorisation**

- (a) A Mongolian PMA authorisation is—

- (1) a design approval issued to the manufacturer of a replacement or modification part that has been found by the Director to comply with the applicable airworthiness requirements; and
  - (2) an authorisation to mark a part with the letters Mongolian PMA.
- (b) No person may identify a part with a Mongolian PMA marking unless that person holds a Mongolian PMA authorisation in respect of that part and the part meets the applicable airworthiness requirements.

#### **21.705. Eligibility**

The applicant for the grant of a Mongolian PMA authorisation shall—

- (1) be the holder of a manufacturing organisation certificate issued under Part 148 and have applied for changes to their exposition for the manufacture of the part; or
- (2) have applied for the grant of a manufacturing organisation certificate issued under Part 148 for the manufacture of the part.

#### **21.707. Application for authorisation**

(a) Each applicant for the grant of a Mongolian PMA authorisation shall complete form CAA 24021/13, which shall require—

- (1) the name and address for service in Mongolia of the applicant; and
- (2) the identity of the product onto which the part is to be installed; and
- (3) a design that consists of—
  - (i) the drawings and specifications necessary to define the configuration and the design features of the part; and
  - (ii) evidence that the configuration and the design features of the part comply with the applicable airworthiness requirements; and
  - (iii) a list of those drawings and specifications specified in paragraph (a)(3)(i); and
  - (iv) the information on dimensions, materials, and processes necessary to define the structural strength of the part; and
- (4) evidence that the applicant can reliably reproduce each part in accordance with the design referred to in paragraph (a)(3); and
- (5) a statement of compliance by the applicant, or for an article that the Director determines includes significant airworthiness requirements, by the holder of a design organisation certificate issued under Part 146, stating that—
  - (i) the design of the part referred to in paragraph (a)(3) is identical to the design of the part that it is to replace or complies with the airworthiness design standards that apply to the product referred to in paragraph (a)(2); and

- (ii) any inspections, tests, and computations required to show that the part complies with the technical conditions of the applicable airworthiness standards have been completed and documented; and
- (6) such further particulars relating to the applicant as may be required by the Director as indicated on the form—

and submit it to the Director with a payment of the applicable fee prescribed by regulations made under the Act.

(b) The applicant shall make the part and associated data available for any inspections that the Director may require.

#### **21.709. Issue of authorisation**

An applicant is entitled to a Mongolian PMA authorisation if—

- (1) the applicant meets the applicable requirements of this Subpart in a manner acceptable to the Director; and
- (2) the granting of the certificate is not contrary to the interests of aviation safety; and
- (3) the applicant has obtained a manufacturing organisation certificate issued under Part 148.

#### **21.711. Privileges of authorisation**

The holder of a Mongolian PMA authorisation for a part may—

- (1) manufacture the part subject to compliance with 21.719; and
- (2) identify the part with the Mongolian PMA marking.

#### **21.713. Validity of authorisation**

The Mongolian PMA authorisation may be suspended or revoked if the Director finds that—

- (1) the conditions required when the authorisation was granted are no longer being observed; or
- (2) the manufacturing organisation certificate issued under Part 148 is suspended or revoked; or
- (3) the responsibilities of the holder specified in 21.719 are no longer being discharged; or
- (4) the article has been shown to give rise to unacceptable hazards in service.

#### **21.715. Duration of authorisation**

(a) A Mongolian PMA authorisation issued under this Subpart shall remain in force until it is suspended or revoked.

(b) The holder of a Mongolian PMA authorisation that is revoked shall forthwith surrender the certificate to the Director.

(c) The holder of a Mongolian PMA authorisation that is suspended shall forthwith produce the certificate to the Director for appropriate endorsement.

#### **21.717. Transfer of authorisation**

A Mongolian PMA authorisation issued under this Subpart is not transferable.

#### **21.719. Responsibilities of authorisation holder**

Each holder of a Mongolian PMA authorisation for a part shall—

- (1) manufacture each part in accordance with the manufacturing organisation certificate issued under Part 148; and
- (2) ensure that each completed part conforms to its design data and is safe for installation; and
- (3) permanently and legibly mark each part with the following information:
  - (i) the information prescribed in 21.813; and
  - (ii) the name, type, or part number of the article; and
  - (iii) the applicable Mongolian PMA number; and
- (4) upon the Director's request, make the records required by 21.721 available to the Director.

#### **21.721. Records**

(a) Each holder of a Mongolian PMA authorisation shall establish procedures to—

- (1) identify, collect, index, store, maintain, and dispose of—
  - (i) records of all parts manufactured by the holder under a Mongolian PMA authorisation; and
  - (ii) all relevant design information, drawings, specifications, tests, analyses, and reports that show the parts comply with the applicable Mongolian PMA and this Part; and
- (2) ensure that—
  - (i) all records are legible and of a permanent nature; and
  - (ii) except as provided in paragraph (b), the records required by paragraph (1) are retained for a period of 2 years from the date the last example of the article is permanently withdrawn from service.



(b) The Director may permit records to be retained for a lesser period than that required by paragraph (a) (2)(ii).

## **Subpart Q — Identification of Products and Parts**

### **21.801. Purpose**

This Subpart prescribes rules governing the identification of—

- (1) aircraft, aircraft engines, and propellers;
- (2) critical parts;
- (3) certain replacement and modification parts.

### **21.803. Identification of aircraft, aircraft engines, and propellers**

(a) Each person who manufactures an aircraft or aircraft engine under Part 148 shall identify the aircraft or engine by means of a fireproof data plate that is—

- (1) marked with the identification information prescribed in 21.805 by etching, stamping, engraving, or other method of fireproof marking; and
- (2) secured in such a manner that it is not likely to be—
  - (i) defaced or removed during normal service; or
  - (ii) lost or destroyed in an accident; and
- (3) for an aircraft other than a manned free balloon, secured to the fuselage so that it is legible to a person on the ground—
  - (i) at an accessible location near an entrance; or
  - (ii) externally on another part of the fuselage; and
- (4) for an aircraft engine, secured to the engine at an accessible location; and
- (5) for a manned free balloon—
  - (i) secured to the balloon envelope; and
  - (ii) located where it is visible to the operator when the balloon is inflated.

(b) Each person who manufactures a manned free balloon shall, in addition to paragraph (a), permanently and legibly mark the basket and any heater assembly with—

- (1) the manufacturer's name; and
- (2) a part number or an equivalent; and

(3) a serial number or an equivalent.

(c) Each person who manufactures a propeller, propeller blade or propeller hub under Part 148 shall identify it by means of a fireproof data plate, or by etching, stamping, engraving, or other method of fireproof marking, that—

- (1) contains the identification information prescribed in 21.805; and
- (2) is placed on a non-critical surface; and
- (3) is not likely to be—
  - (i) defaced or removed during normal service; or
  - (ii) lost or destroyed in an accident.

### **21.805. Identification information**

The identification information that is required to be marked on the data plate under rule 21.803(a) (1), and for the identification of a propeller, propeller blade, or propeller hub under rule 21.803(c)(1) must include—

- (1) the manufacturer's name; and
- (2) the model designation; and
- (3) the manufacturer's serial number; and
- (4) if applicable, the type certificate or type acceptance certificate number; and
- (5) if applicable, the manufacturing organisation certificate number or foreign equivalent; and
- (6) for an aircraft engine, the established rating; and
- (7) any other information that the Director may require.

### **21.807. Removal, alteration and replacement of identification information**

(a) Except as provided in paragraph (b), a person must not remove, alter, or replace any of the following without the approval of the Director—

- (1) the identification information that is required under rule 21.803(a)(1) to be marked on a data plate;
- (2) the identification information that is required under rule 21.803(c)(1) to identify a propeller, propeller blade, or propeller hub;
- (3) the part number and serial number that is required under rule 21.811 for the identification of a critical part.

(b) A person performing maintenance in accordance with Part 43 may remove, alter, or replace the identification information referred to in paragraphs (a)(1) and (a)(2) and the part and serial number referred to in paragraph (a)(3) if the removal, alteration, or replacement is carried out in accordance with a method, technique, or practice that is acceptable to the Director.

**21.809. Removal and reinstallation of data plate**

(a) Except as provided by paragraph (b), a person shall not remove or reinstall the data plate containing the identification information prescribed in 21.805 without the approval of the Director.

(b) A person performing maintenance in accordance with Part 43 may remove or reinstall the data plate containing the identification information prescribed in 21.805 if—

- (1) the removal of the data plate is necessary during the maintenance; and
- (2) the data plate is removed and reinstalled in accordance with methods, techniques, and practices acceptable to the Director; and
- (3) the removed data plate is reinstalled on the product or part from which it was removed.

**21.811. Identification of critical parts**

Each person who manufactures a critical part shall permanently and legibly mark the part with—

- (1) a part number or an equivalent; and
- (2) a serial number or an equivalent.

**21.813. Identification of replacement and modification materials, parts, and appliances**

(a) Except as provided in paragraph (b), each person who manufactures a replacement or modification part or appliance under an authorisation required by Subpart K shall, in addition to the identification information prescribed in 21.805, permanently and legibly mark the part or appliance with—

- (1) the letters 'Mongolian TSO' or 'Mongolian PMA' as appropriate; and
- (2) the name, trademark, or symbol of the holder of the authorisation; and
- (3) the part number; and
- (4) the name and model designation of each product issued with a type certificate or type acceptance certificate, on which the part is eligible for installation.

(b) Each person who manufactures a replacement or modification material, part, or appliance under 21.303(2) shall permanently and legibly mark the material, part, or appliance in such a manner as to ensure it can be—

- (1) identified separately to those otherwise acceptable materials, parts, and appliances; and

(2) clearly related to its manufacturing data.

(c) If a material, part, or appliance is too small or it is otherwise impractical to mark the material, part, or appliance with the information required by paragraphs (a) or (b), the information shall be recorded on a tag attached to the material, part, appliance, or its container.

(d) Where the marking required by paragraph (a)(4) is so extensive that to record it on a tag is impractical, the tag attached to the material, part, appliance, or the container may refer to a specific readily available manual or catalogue for the name and model designation of each product issued with a type certificate or type acceptance certificate, on which the material, part, or appliance is eligible for installation.

## Appendix A — N/A

## Appendix B — Reserved

## Appendix C

### Airworthiness Design Standards

(a) Subject to paragraph (c), the standard category airworthiness design standards are—

- (1) any of the following Federal Aviation Regulations issued by the Federal Aviation Administration of the United States of America:
  - (i) Part 23 — Airworthiness Standards: Normal, Utility, Acrobatic, and Commuter Category Aeroplanes, excluding 23.785(a) and 23.807(b)(5):
  - (ii) Part 25 — Airworthiness Standards: Transport Category Aeroplanes:
  - (iii) Part 27 — Airworthiness Standards: Normal Category Rotorcraft:
  - (iv) Part 29 — Airworthiness Standards: Transport Category Rotorcraft:
  - (v) Part 31 — Airworthiness Standards: Manned Free Balloons:
  - (vi) Part 33 — Airworthiness Standards: Aircraft Engines:
  - (vii) Part 35 — Airworthiness Standards: Propellers; or
- (2) a set of airworthiness design standards that the Director determines—
  - (i) comply with ICAO Annex 8; and
  - (ii) provide an equivalent level of safety to those airworthiness design standards specified in paragraph (a)(1).

(b) Subject to paragraph (c), the restricted category airworthiness design standards are—

- (1) any of the design standards specified in paragraph (a), and any relevant standards specified in paragraphs (d) and (e), excluding those requirements that the Director determines inappropriate for the purpose for which the aircraft is to be used; or
- (2) a set of airworthiness design standards that the Director determines appropriate for the purpose for which the aircraft is to be used.

(c) The airworthiness design standard for a standard category or a restricted category aircraft must include a flight manual that contains—

- (1) the operating limitations and information required to be provided by the applicable airworthiness design standard, in the form of a manual, markings, or placards; and

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- (2) for an aeroplane and rotorcraft, the maximum ambient atmospheric temperatures for which engine cooling was demonstrated, in the performance information section of the flight manual; and
  - (3) the information required under rule 21.32 in the form of a certificate or a page in the flight manual.
- (d) The standard category aircraft noise standards are—
- (1) the standards specified in the applicable chapter of ICAO Annex 16 Volume I except that —
    - (i) after 30 June 2003 every subsonic turbo-jet and turbo-fan powered aeroplanes must comply with the standards specified in Chapter 3 of Annex 16 Volume I; and
    - (ii) if a higher standard has been specified in the aircraft type certificate, noise certificate or equivalent document, then that standard applies; or
  - (2) a set of equivalent airworthiness design standards acceptable to the Director.
- (e) The standard category aircraft engine emission standards are—
- (1) the standards specified in the applicable chapter of ICAO Annex 16 Volume II; or
  - (2) a set of equivalent airworthiness design standards acceptable to the Director.

## Appendix D

### Acceptable Technical Data

- (a) Subject to paragraph (b), the following are acceptable technical data:
- (1) a type certificate data sheet:
  - (2) a foreign type certificate data sheet used for the issue of a type acceptance certificate:
  - (3) type design data for a type certificated product:
  - (4) design change data that supports a design change approved by the means specified in rule 21.73:
  - (5) data approved by the Director under rule 21.505:
  - (6) data provided by the Director in an Advisory Circular:
  - (7) an airworthiness directive that gives a specific instruction for modification or repair:
  - (8) a supplemental type certificate issued by the following:
    - (i) the Federal Aviation Administration of the United States of America:

- (ii) the Federal Aviation Safety Authority of Russia:
  - (iii) EASA of the EU:
  - (9) N/A:
  - (10) data giving a specific instruction for modification or repair contained in a maintenance manual, repair manual, overhaul manual, instruction for continued airworthiness, service bulletin, or an equivalent provided by the manufacturer of the product for which it is to be used and which is listed in the type certificate, or by reference in the type acceptance certificate:
  - (11) AC 43.13-1B, issued by the Federal Aviation Administration of the United States of America:
  - (12) data included in and specific to the category of an airworthiness certificate.
- (b) The technical data listed in paragraph (a) are acceptable if—
- (1) the data is appropriate to the product, component, or appliance, and is directly applicable to the work being carried out; and
  - (2) for a foreign supplemental type certificate or supplemental type approval—
    - (i) a complete new flight manual is not introduced; and
    - (ii) the aircraft type is not re-designated; and
    - (iii) the data is supplemental to the particular type certificate accepted by the Director and that type certificate is referenced on the supplemental type certificate or supplemental type approval; and
  - (3) the installer has the written permission of the holder of the supplemental type certificate or supplemental type approval to install the STC; and
  - (4) data provided by the manufacturer of a component does not conflict with data provided by the manufacturer of the product or assembly of which the component is to form a part.

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